

# GENERATION OF TURBULENT INLET VELOCITY CONDITIONS FOR LARGE EDDY SIMULATIONS

Hugo G. Castro and Rodrigo R. Paz

CONICET  
Centro Internacional de Métodos Computacionales en Ingeniería  
CIMEC-INTEC-UNL  
UTN-FRRe  
Argentina  
<castrohgui@gmail.com>

10th World Congress on Computational Mechanics



10<sup>th</sup> World Congress on  
Computational Mechanics  
© 19 July 2012 • São Paulo • Brazil

## Intro and General Objectives

- ▷ Large eddy simulation (LES) has become an attractive approach to solve fluid flow problems due to the improvement of computational power.
- ▷ In order to obtain a fully developed turbulence flow without a significant increase in computational costs one must assure that the turbulence inlet conditions are adequately prescribed.
- ▷ Several methods are available (Tabor and Baba-Ahmadi, *Computer and Fluids*, 2010):
  - ▷ Precursor simulation methods: *cyclic domains, prepared library, concurrent library generation.*
  - ▷ Synthesis method: *Fourier techniques, digital filter based method, proper orthogonal decomposition analysis.*



## Intro and General Objectives

Objective of this research:

- ▷ Develop a synthesis turbulence method to be used in the generation of inflow boundary conditions for LES with application to turbulent fluid flows.
- ▷ Key points of the proposed technique:
  - ▷ Statistical properties of turbulence.
  - ▷ Isotropic and anisotropic turbulence.
  - ▷ Spectral target function.

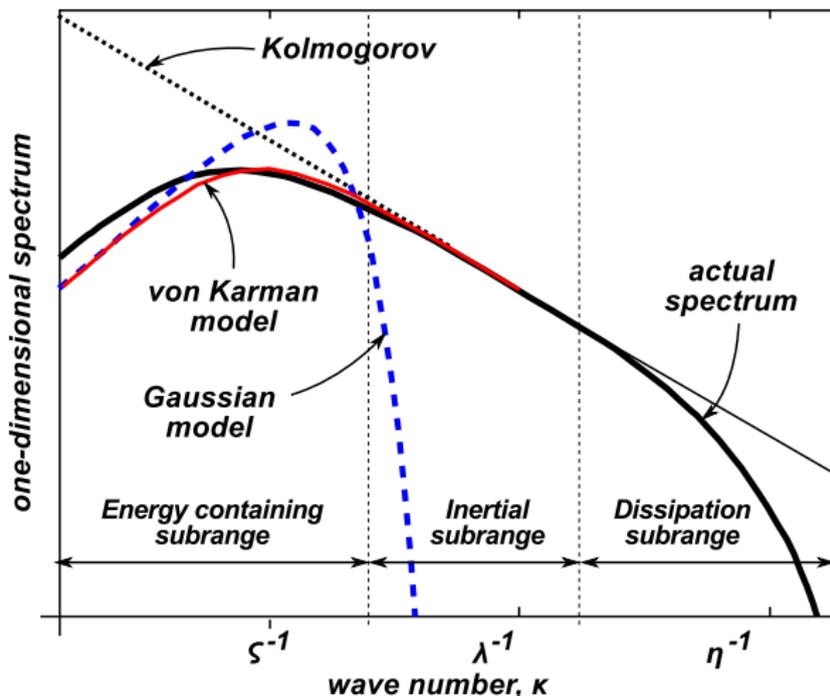
# Modified discretizing and synthesizing random flow generation (MDSRFG)

- ▷ The proposed methodology is based on previous approaches:
  - ▷ Random flow generation technique (RFG, Smirnov et al, *Journal of Fluids Engineering*, 2001):
    - Inhomogeneous and anisotropic turbulence flow.
    - Time and spatial scales incorporated in the formulation.
    - Included in the software FLUENT (spectral synthesizer).
    - Gaussian's spectral model only.
  - ▷ Discretizing and synthesizing random flow generation method (DSRFG, Huang et al, *Journal of Wind Engineering and Industrial Aerodynamics*, 2010):
    - Any model spectrum can be used.
    - Inhomogeneous and anisotropic turbulence flow.
    - Highly parallelizable algorithmic implementation.
    - Time correlation?.
    - What about frequency interval size ( $\Delta f$ )?.



## MDSRFG (cont.)

Huang et al. (2010) analysis:



## MDSRFG (cont.)

According to Huang et al. (2010) (based on the work of Smirnov et al. (2001) and Kraichnan (1970)) an inhomogeneous and anisotropic turbulent flow field  $\mathbf{u}(\mathbf{x}, t)$  can be synthesized as follows:

$$u_i(\mathbf{x}, t) = \sum_{m=1}^M \sum_{n=1}^N \left[ p_i^{m,n} \cos \left( \tilde{k}_j^{m,n} \tilde{x}_j + \omega_{m,n} t \right) + q_i^{m,n} \sin \left( \tilde{k}_j^{m,n} \tilde{x}_j + \omega_{m,n} t \right) \right],$$

where  $M$  and  $N$  are the number of wavenumbers ( $k_m$ ) considered in the discretization of the target spectrum and the sample size of each  $k_m$ , respectively, and  $\omega_{m,n} \in N(0, k_m U_{\text{avg}})$ .



## MDSRFG (cont.)

and

$$p_i^{m,n} = \text{sign}(r_i^{m,n}) \sqrt{\frac{4}{N} E_i(k_m) \frac{(r_i^{m,n})^2}{1 + (r_i^{m,n})^2}},$$

$$q_i^{m,n} = \text{sign}(r_i^{m,n}) \sqrt{\frac{4}{N} E_i(k_m) \frac{1}{1 + (r_i^{m,n})^2}}.$$

$$\tilde{x} = \frac{x}{L_s},$$

$$\tilde{k}^{m,n} = \frac{k^{m,n}}{k_0},$$

$$L_s = \theta_1 \sqrt{L_u^2 + L_v^2 + L_w^2}.$$

where  $r_i^{m,n} \in N(0, 1)$ ,  $\theta_1$  is a scalar value between 1 and 2 used to adjust the spatial correlation.

## MDSRFG (cont.)

$$p_i^{m,n} = \text{sign}(r_i^{m,n}) \sqrt{\frac{4}{N} E_i(k_m) \frac{(r_i^{m,n})^2}{1 + (r_i^{m,n})^2}}$$

$$q_i^{m,n} = \text{sign}(r_i^{m,n}) \sqrt{\frac{4}{N} E_i(k_m) \frac{1}{1 + (r_i^{m,n})^2}}$$

these factors align the energy spectrum according to the anisotropy conditions of the turbulence, providing a synthesized velocity series that must satisfy the mean square values in each of the spatial coordinate axes.



## MDSRFG (cont.)

Starting from the mean square value of a random function  $f(t)$  definition:

$$f_{\text{rms}}^2(t) = \lim_{T \rightarrow \infty} \frac{1}{T} \int_0^T f^2(t) dt$$

we obtain:

$$u_{\text{rms},i}^2(x, t) = \lim_{T \rightarrow \infty} \frac{1}{T} \int_0^T \left\{ \sum_{m=1}^M \sum_{n=1}^N [p_i^{m,n} \cos(\tilde{k}_j^{m,n} \tilde{x}_j + \omega_{m,n} t) + q_i^{m,n} \sin(\tilde{k}_j^{m,n} \tilde{x}_j + \omega_{m,n} t)] \right\}^2 dt.$$

## MDSRFG (cont.)

And after some mathematical manipulations:

$$\begin{aligned}\overline{u_i u_i} &= \frac{1}{2} \sum_{m=1}^M \sum_{n=1}^N p_i^{m,n} p_i^{m,n} + \frac{1}{2} \sum_{m=1}^M \sum_{n=1}^N q_i^{m,n} q_i^{m,n} \\ &= 2 \int_0^\infty E(k) dk \approx 2 \sum_{m=1}^M E(k_m) \Delta k_m,\end{aligned}$$

hence,

$$\begin{aligned}\overline{u_i u_i} &= \frac{1}{2} \sum_{m=1}^M \sum_{n=1}^N \sum_{i=1}^3 \left[ \frac{4}{N} E_i(k_m) \frac{(r_i^{m,n})^2}{1 + (r_i^{m,n})^2} + \frac{4}{N} E_i(k_m) \frac{1}{1 + (r_i^{m,n})^2} \right] \\ &= \frac{2}{N} \sum_{m=1}^M \sum_{n=1}^N E(k_m) = 2 \sum_{m=1}^M E(k_m),\end{aligned}$$

## MDSRFG (cont.)

$$\overline{u_i u_i} = 2 \sum_{m=1}^M E(k_m)$$

as  $E(k_m)$  is a positive quantity for any  $k$ , the kinetic energy is represented by a divergent series. This causes a strong dependency of the turbulence intensity of the generated fluctuating velocities with the number of points  $M$  considered to discretize the target spectrum.

CONICET



10<sup>th</sup> World Congress on  
Computational Mechanics

© 19 July 2012 • São Paulo • Brazil

## MDSRFG (cont.)

So, some modifications to the DSRFG method are proposed:

$$u_i(\mathbf{x}, t) = \sum_{m=1}^M \sum_{n=1}^N \left[ p_i^{m,n} \cos \left( \tilde{k}_j^{m,n} \tilde{x}_j + \omega_{m,n} \frac{t}{\tau_0} \right) + q_i^{m,n} \sin \left( \tilde{k}_j^{m,n} \tilde{x}_j + \omega_{m,n} \frac{t}{\tau_0} \right) \right],$$

where  $p_i^{m,n}$  and  $q_i^{m,n}$  are modified according to the previous analysis and  $\tau_0$  is added in order to handle the time correlation of the series.

$$p_i^{m,n} = \text{sign}(r_i^{m,n}) \sqrt{\frac{4c_i}{N} E_i(k_m) \Delta k_m \frac{(r_i^{m,n})^2}{1 + (r_i^{m,n})^2}},$$

$$q_i^{m,n} = \text{sign}(r_i^{m,n}) \sqrt{\frac{4c_i}{N} E_i(k_m) \Delta k_m \frac{1}{1 + (r_i^{m,n})^2}}.$$

## Validation of the procedure

1) Same test as in the work of Huang et al., *Journal of Wind Engineering and Industrial Aerodynamics*, 2010): an inhomogeneous and anisotropic turbulent flow field.

▷ von Kármán models:

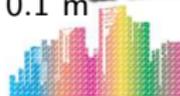
$$S_u(f) = \frac{4(I_u U_{\text{avg}})^2 (L_u / U_{\text{avg}})}{[1 + 70.8(fL_u / U_{\text{avg}})^2]^{5/6}},$$

$$S_v(f) = \frac{4(I_v U_{\text{avg}})^2 (L_v / U_{\text{avg}}) (1 + 188.4(2fL_v / U_{\text{avg}})^2)}{[1 + 70.8(2fL_v / U_{\text{avg}})^2]^{11/6}},$$

$$S_w(f) = \frac{4(I_w U_{\text{avg}})^2 (L_w / U_{\text{avg}}) (1 + 188.4(2fL_w / U_{\text{avg}})^2)}{[1 + 70.8(2fL_w / U_{\text{avg}})^2]^{11/6}}.$$

▷ Turbulence intensity values:  $I_u = 8\%$ ,  $I_v = 16\%$ ,  $I_w = 24\%$ .

▷ Turbulence integral length scales:  $L_u = 0.6$  m,  $L_v = 0.3$  m,  $L_w = 0.1$  m



## Validation of the procedure (cont.)

To apply the procedure we must first obtain the  $c_i$  values for  $p_i$  and  $q_i$ :

$$u_{\text{rms},1}^2 = (I_u U_{\text{avg}})^2 = 2c_1 \int_0^\infty S_u(k) dk$$

$$\approx 2c_1 \cdot 0.2377 \beta\left(\frac{1}{3}, \frac{1}{2}\right) I_u^2 U_{\text{avg}} \Rightarrow c_1 = \frac{U_{\text{avg}}}{2},$$

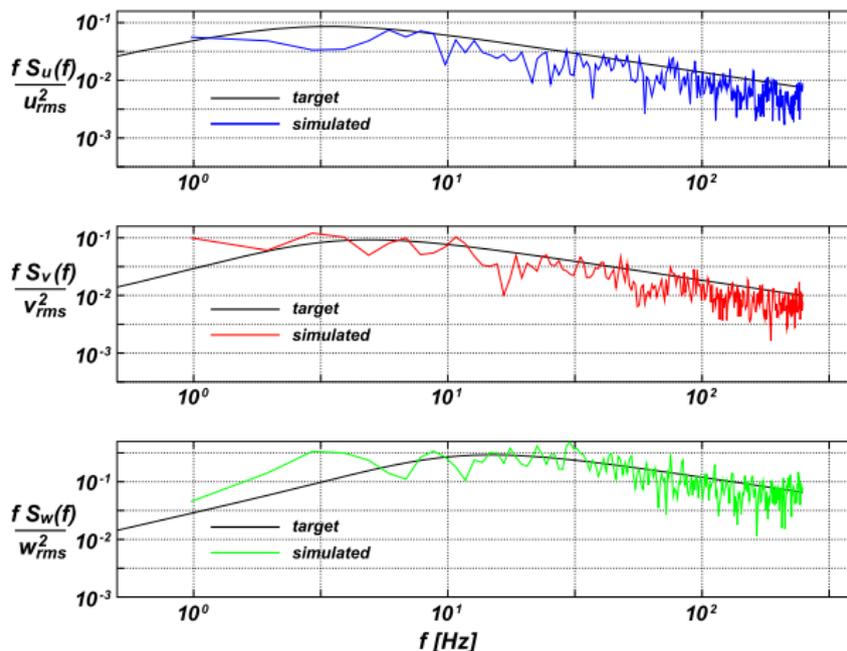
$$u_{\text{rms},2}^2 = (I_v U_{\text{avg}})^2 = 2c_2 \int_0^\infty S_v(k) dk$$

$$\approx 2c_2 \left[ 0.1189 \beta\left(\frac{1}{2}, \frac{4}{3}\right) + 0.3163 \beta\left(\frac{1}{3}, \frac{3}{2}\right) \right] I_v^2 U_{\text{avg}}$$

$$\Rightarrow c_2 = c_3 \approx \frac{U_{\text{avg}}}{2}.$$

## Validation of the procedure (cont.)

Comparison of the spectra by the MDSRFG method and the target spectrum:



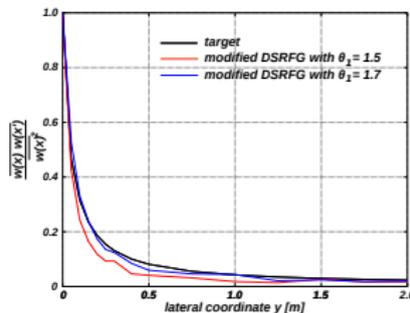
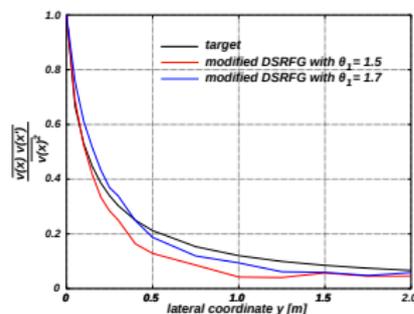
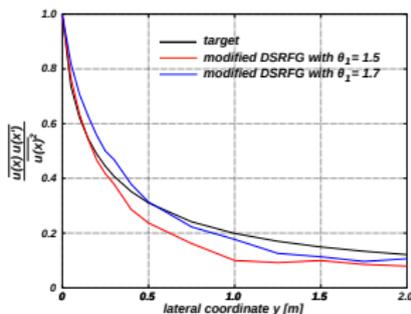
## Validation of the procedure (cont.)

Comparison of the rms values of the simulated fluctuating velocities:

	$\sigma_u$	$\sigma_v$	$\sigma_w$
Scaling and transformation	0.9968	2.44	2.9956
Aligning and remapping	0.95	1.9987	3.08
MDSRFG approach	1.0527	2.1850	3.1123
target	<b>1.12</b>	<b>2.24</b>	<b>3.36</b>

# Validation of the procedure (cont.)

Spatial correlation:



## Validation of the procedure (cont.)

Time correlation: samples of temporal correlations from the MDSRFG and the DSRFG methods are shown in the next figures for each velocity component. Also, they are compared to the autocorrelation function of a random stationary process:

$$R_i(\tau) = e^{-|\tau|/T_i},$$

with time scales  $T_i$  ( $i = u, v, w$ ) computed as

$$T_i = \int_0^{\infty} R_i(\tau) d\tau \equiv \sum_{j=0}^{M_0} R_i(j \delta\tau) \delta\tau, \quad (1)$$

where  $M_0 < M$ . Low frequency fluctuations cause oscillations on the time correlation around the zero value as the time lag tends to infinite. Consequently, if equation (1) is approximated without an adequate upper limit of the sum, it will fail to estimate the scale. In this work the time scale is computed by setting  $M_0$  to the first  $\tau$ -axis crossing value.



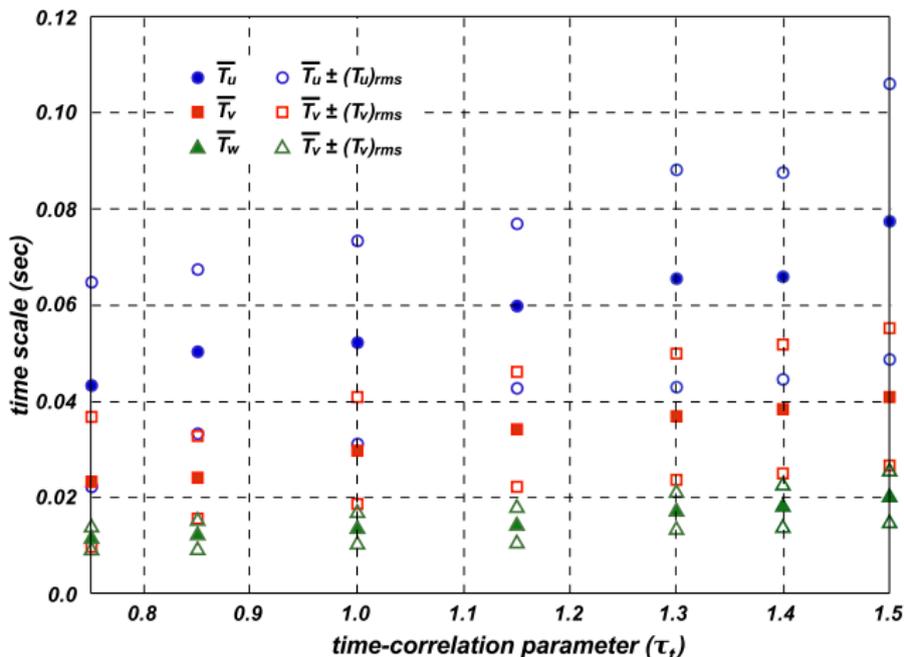
## Validation of the procedure (cont.)

Time correlation: Time scale statistics comparison (sec).

	$T_u$	$T_v$	$T_w$
DSRFG approach	$0.034 \pm 0.028$	$0.022 \pm 0.009$	$0.010 \pm 0.002$
MDSRFG approach	$0.043 \pm 0.021$	$0.023 \pm 0.014$	$0.011 \pm 0.002$
target	0.043	0.021	0.007

## Validation of the procedure (cont.)

Time correlation: Time scale statistics of the fluctuating velocity components as a function of  $\tau_0$  obtained by the MDSRFG method.



## Some remarks

- ▷ For each node at the inlet section the cost at each time step is  $O(MN)$ . (Same as in the DSRFG method)
- ▷ The turbulence synthesis for some number of time steps (or the entire simulation process) can be done prior to the LES computations. (Same as in the DSRFG method)
- ▷ The anisotropic turbulence conditions at the inlet plane can be obtained by performing a previous RANS simulation or by experimental measurements. The two input parameters,  $L_s$  and  $\tau_0$  must be selected in order to reproduce the statistical properties of the flow under consideration. (Same as in the DSRFG method in the case of  $L_s$ )



## Some remarks

- ▷ Is important to highlight the possibility to slightly modify the time scale with different  $\tau_0$  values in the MDSRFG method while the DSRFG method are limited in this way. Note that even the target time scales, estimated by the Taylor's hypothesis, are in accordance with the values obtained by the DSRFG method, the application of the MDSRFG approach leads to a wider range of possible values by changing  $\tau_0$ .
- ▷ Also, to validate the relation proposed for the equations of  $p_i$  and  $q_i$ , the influence of the frequency interval size  $\Delta f$  over the rms values of the time series was analyzed. What it is expected is that the synthetic turbulence generation provides a correct rms values as  $\Delta f \rightarrow 0$ , that is, as the discretization of the spectrum becomes finer the energy turbulence content in each frequency will be included in the time series generation.

## Some remarks

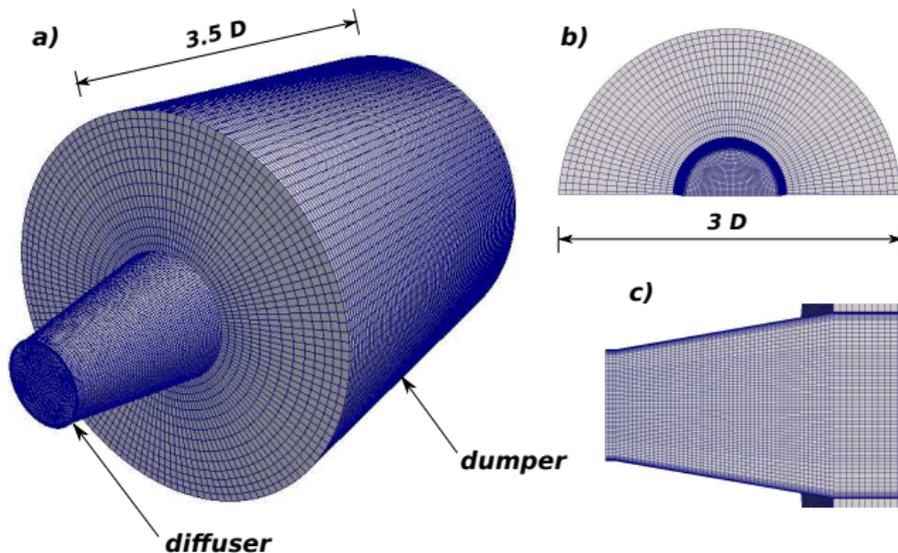
It can be seen that the method proposed in this work converges to the target values as  $\Delta f$  becomes smaller while in the case of the DSRFG method the values do not converge at all.

$\Delta f$	$\sigma_u$		$\sigma_v$		$\sigma_w$	
	DSRFG	MDSRFG	DSRFG	MDSRFG	DSRFG	MDSRFG
10	0.49	0.76	1.27	1.84	2.05	2.95
5	0.83	0.87	1.93	2.07	2.94	3.01
2	1.47	0.98	3.15	2.11	4.69	3.03
1	2.17	1.04	4.47	2.11	6.75	3.11
target	1.12		2.24		3.36	

▷ As a final observation we notice that the proposed approach, as any synthesized turbulence generation method, must be used as a turbulence initializer, i.e., a perturbation generator that “triggers” the transition to a fully developed turbulence state by LES. In this regard, it must be said that independently of the selected  $L_s$  value, the resolved scales are in concordance with the mesh (filter) size which is inherent to the LES conception.

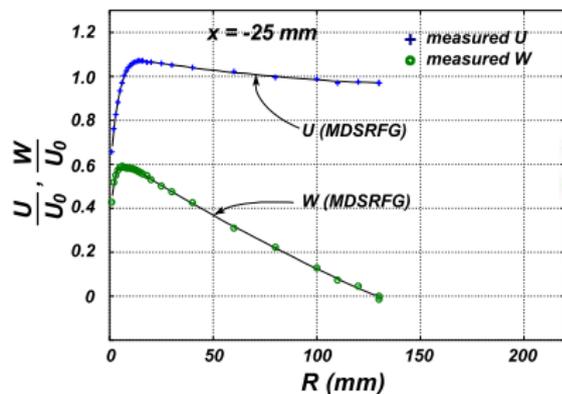
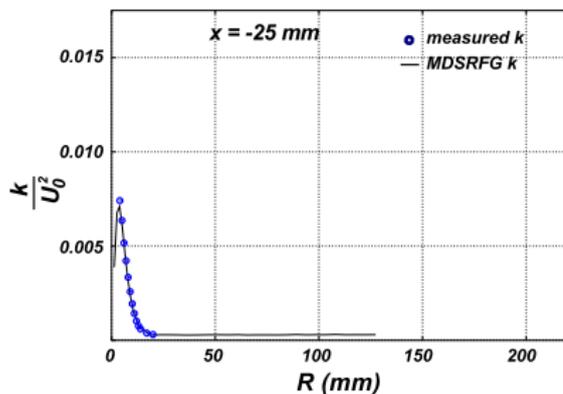
## Validation of the procedure (2)

2) This test case consists of a swirling boundary layer developing in a conical diffuser that was experimentally studied by Clausen *et al* (1993).



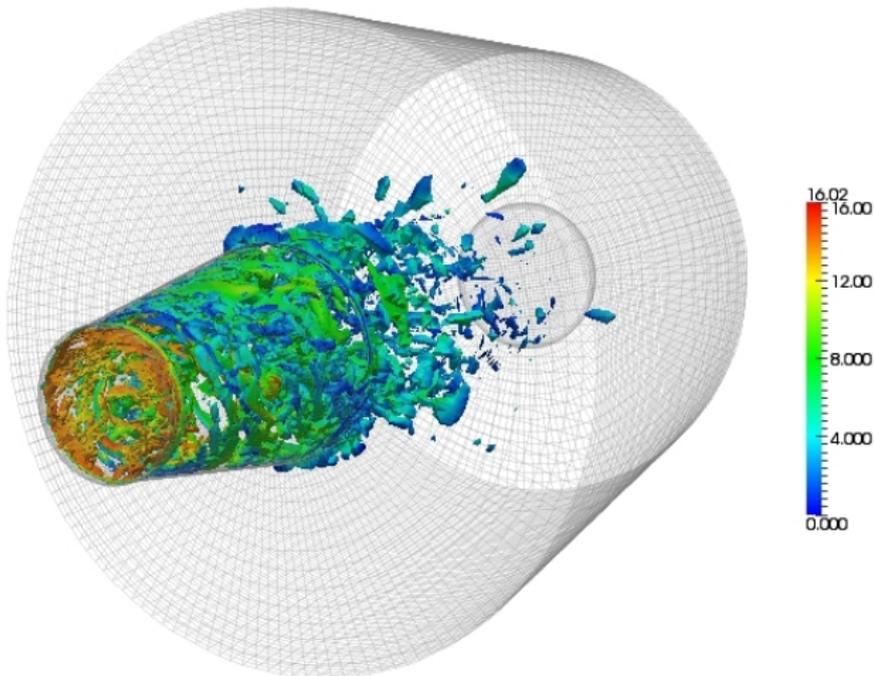
## Validation of the procedure (2)

Inlet conditions by the MDSRFG method:



## Validation of the procedure (2)

Q-vortex structures:



CONICET

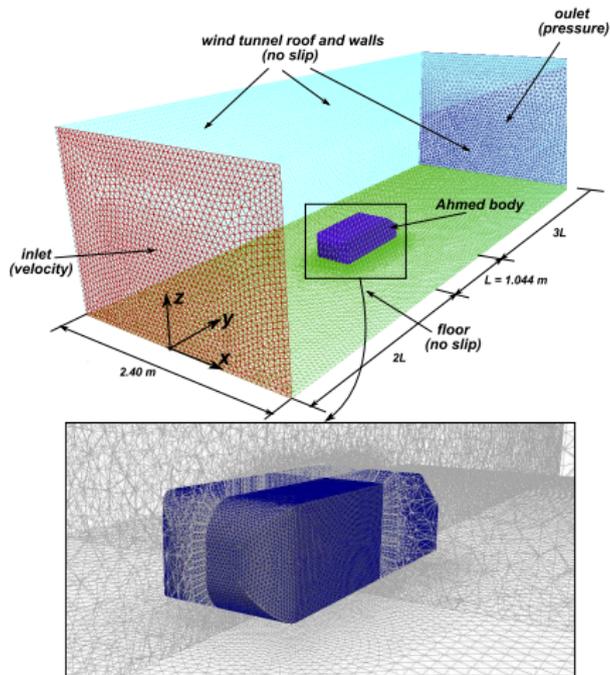


10<sup>th</sup> World Congress on  
Computational Mechanics

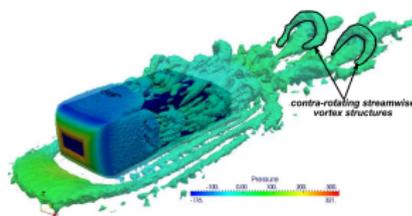
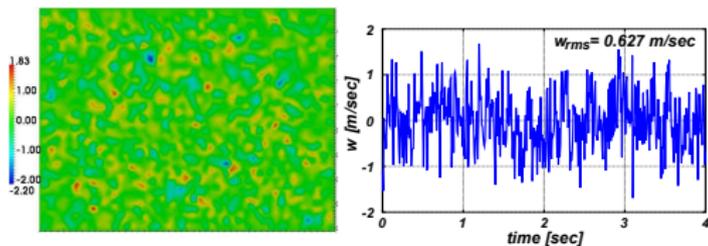
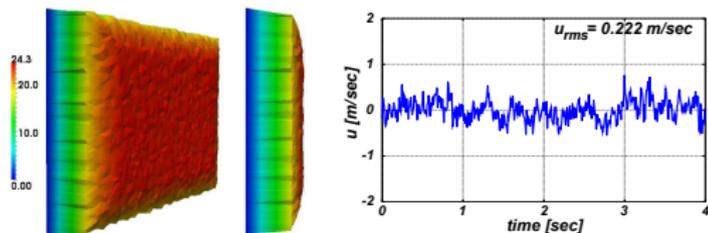
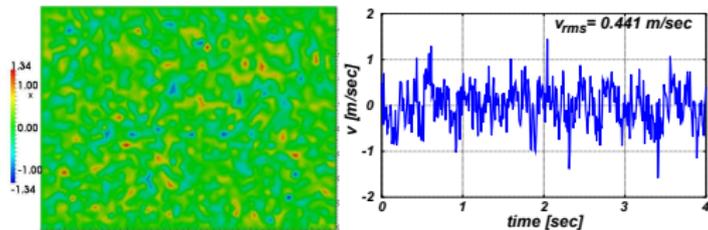
© 19 July 2012 • São Paulo • Brazil

## Validation of the procedure (3)

### 3) Simulation of a wind tunnel inlet flow field (Ahmed body test).



# Validation of the procedure (3)



CONICET

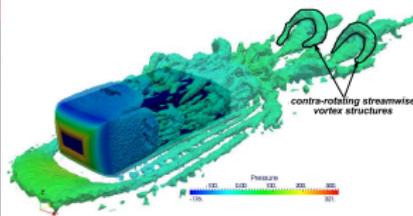
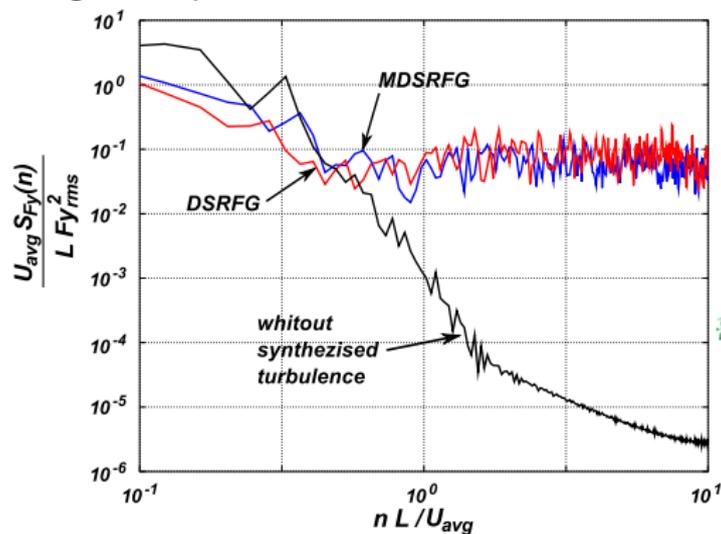


10<sup>th</sup> World Congress on  
Computational Mechanics

© 13 July 2012 • São Paulo • Brazil

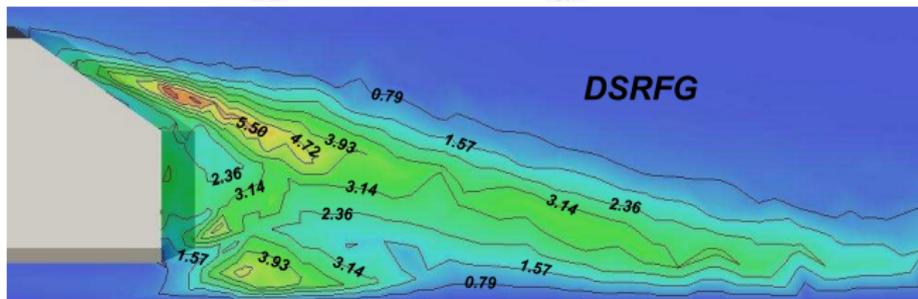
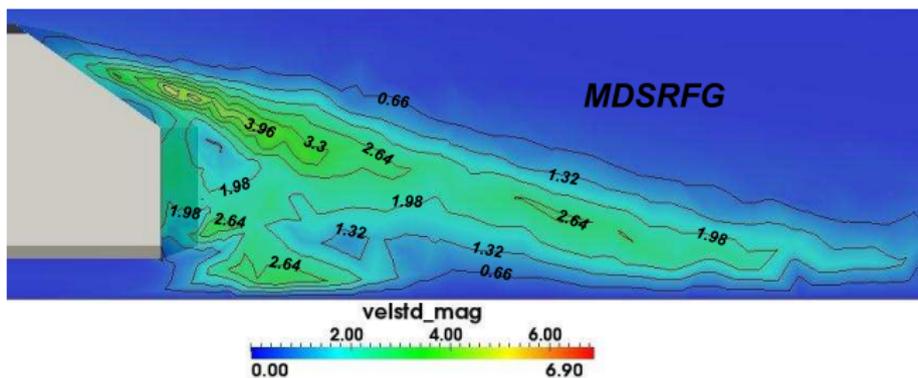
## Validation of the procedure (3)

Drag force spectrum.



## Validation of the procedure (3)

Standard deviation velocity magnitude.



## Conclusions

- ▷ A general method for the generation of inflow synthesized turbulence was presented and evaluated. The method is based on the DSRFG method, preserving its main characteristics and advantages.
- ▷ The key point of the MDSRFG method presented in this study is that it preserves the statistical quantities that would be prescribed at the inlet of the domain independently of the number of samples  $M$  (number of points in the spectrum) considering in the computation of the factors  $p_i^{m,n}$  and  $q_i^{m,n}$ .
- ▷ As each fluctuating velocity component is generated in each node independently of the others, the method is highly parallelizable. Furthermore, the generation of each nodal fluctuating velocity component can be done previously to the computation by LES, calling in each time step the corresponding nodal value.



# Conclusions

- ▷ Analysing the swirling flow inside a diffuser problem, the MDSRFG method has shown that can set properly the inlet conditions for LES of turbulent flows.
- ▷ With regard to the simulation of the flow over the Ahmed body, it has been shown that the results obtained in the LES, i.e., force coefficients, level of unsteadiness in the wake and the back-light of the model, are very sensitive to the upstream inflow conditions.

CONICET



10<sup>th</sup> World Congress on  
Computational Mechanics

© 19 July 2012 • São Paulo • Brazil

